

The SAR-C Payload Designed for µsat constellations

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DRAGONFLY

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NEOS NATIONAL EARTH OBSERVATIONS AND SPACE SECRETARIAT



## Introduction

Commercial SAR satellite constellations have grown since 2017 and are having a drastic impact on the earth observation landscape

SAR satellites are the only 24/7 sensors – imaging day and night in ALL weather

Constellations has reduced revisit rates to be daily or every few hours hourly

Resolutions achievable in commercial satellites are as fine as 25cm

Unprecedented capability to image the earth for a myriad of applications

However, requirements for imaging still far exceed the capability of current satellite constellations

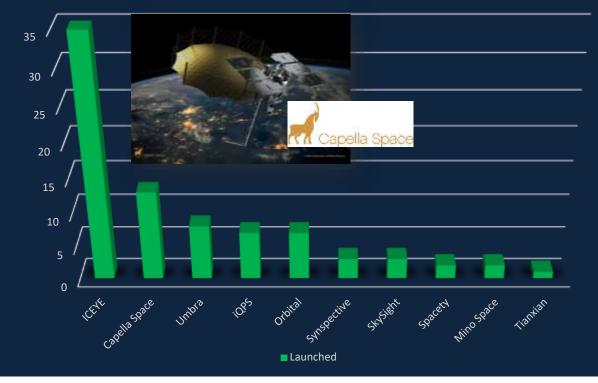
- •SAR payloads operate typically < 5% of orbit
- •Most of these satellites do not provide polarimetric information
- •Almost all commercial satellites are at X-band (not suited to certain applications)

Additionally, recent reductions in launch costs allows for more complexity in the satellite with optimal performance vs mass likely in the 100 – 200 kg range

- •Onboard processing to reduce data rates
- More advanced modes

## ICEYE

#### # of SAR satellites launched to date





## Introduction

A **C-band polarimetric SAR constellation** could provide significant added value to the global SAR market



#### **C-Band Payload Design Requirements**

Payload Mass	Ideally < 150 kg
Payload Power	Ideally < 2 kW average during operation
Frequency	5.25 to 5.85 GHz
Resolution	<1m in high res
Polarisation	Ideally Quad Pol (VV,VH,HV,HH)
Modes	Wide Swath for Maritime Surveillance (ideally with onboard detection)
	High Resolution for Security and Infrastructure Monitoring
Cost	Affordable for Developing Countries
Design	Easy to manufacture in high numbers



## **The SAR-C Payload**

#### 1m

## A flexible design for C-band µsat SAR missions

<b>Operational Lifetime</b>	5 years, Payload and Avionics fully redundant							
Operating Band	C-band (5.5 GHz)							
Bandwidth	600 MHz (0.25m range resolution)							
Polarisation	Quad Pol (VV, VH and HV, HH pulse to pulse)							
Peak Transmit Power	Up to 16 kW (up to 18% duty cycle)							
Imaging Time per Orbit	Up to 5 min							
Imaging Modes (Resolution / Swath)	Spotlight	Sliding Spotlight	Stripmap	Stripmap Wide	ScanSAR 100	ScanSAR 300		
	0.5 – 1.5 m	1-2 m	3 m	5 m	10 - 15 m	25 -30 m		
	10 x 10 km	20x20 km	20 – 44 km	60 - 80 km	100 km	300 km		
Extended Imaging Modes to allow further R&D	VideoSAR, Onboard and Real-time Processing Modes, Moving Target Detection, Multi-Input Multi-Output (MIMO) R&D Modes							
Mass	Payload				~170 kg			
Launch	Foldable to typical ride-share launch volumes							
Interfaces	Standard interfaces to data recorders and mission computers							
Onboard Processing	(design goal) Parallel output datapath that allows processing whilst still recording							

Rideshare launch depiction on SpaceX

#### Antenna size driven by

- Required gain (~42 dBi Antenna Area)
- Doppler constraints (Antenna Length)
- Swath requirements (Antenna Height)
- Power requirements (Antenna Area)
- Launch Volume

5.5m



Electronic beam steering allows for fast swath switching and left/right imaging

Multi phase-center antenna with digital beamforming allows for HRWS and ATI modes

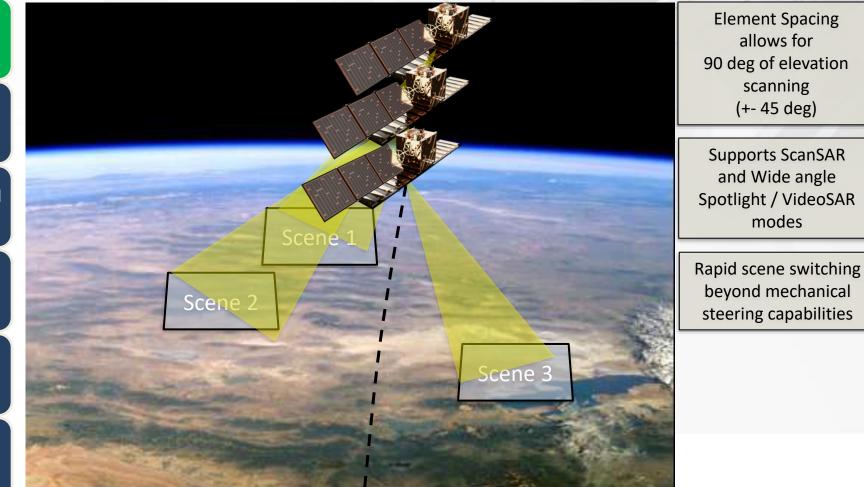
Antenna front end designed to support dual/quad polarisation

Based largely on proven technology that has been flown in airborne tests and which is used in other radar designs

Designed for production in volume as is required for constellations

Redundancy to align with a 5 year lifetime

Digital architecture allows for evaluation of onboard processing in tech-demo phase





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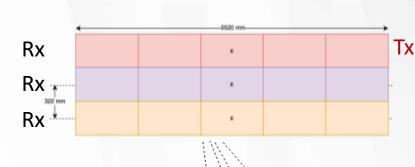
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- V Similar performance with shorter antenna
- Tx Redundancy

Ground Range

TX-swath

- Wider along track beam coverage
- Finer achievable cross range resolution
- Improved multi-looking at same resolution

RX-DBF-swath #1

RX-DBF-swath #2

RX-DBF-swath #3

Multiple Rx beams along track Along track interferometry Stripmap modes with resolution < L/2Widebeam Spotlight modes

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Flight Track

Wider Azimuth (Doppler)



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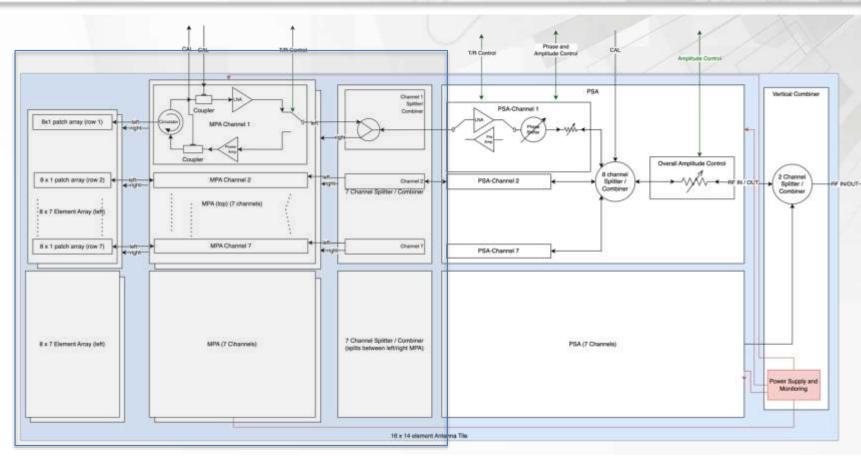
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Dual Channel Design in Tx / RX Front End provides for easier path to upgrade to Dual or Quad Polarisation



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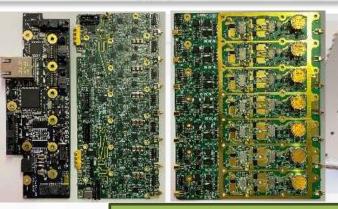
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Phase array front ends

C-Band multi-channel Receiver/Exciter

Airborne multi-channel array antenna

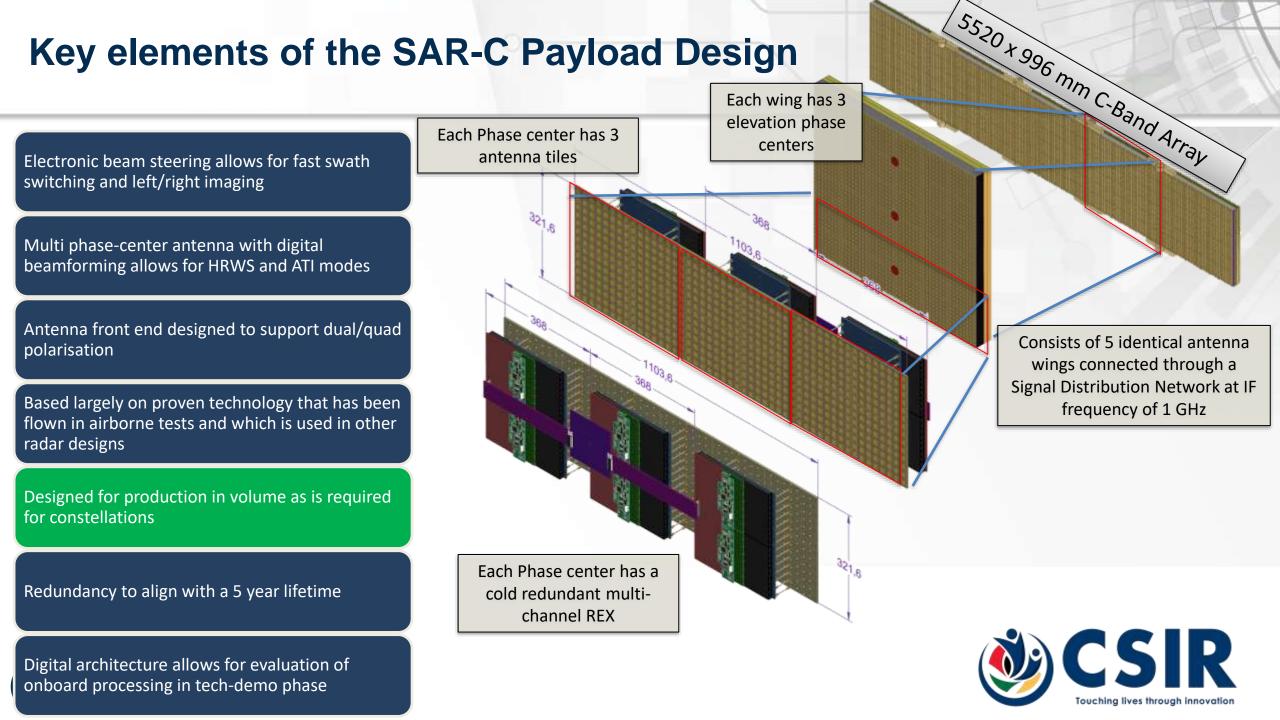
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**Real-time Processor** 

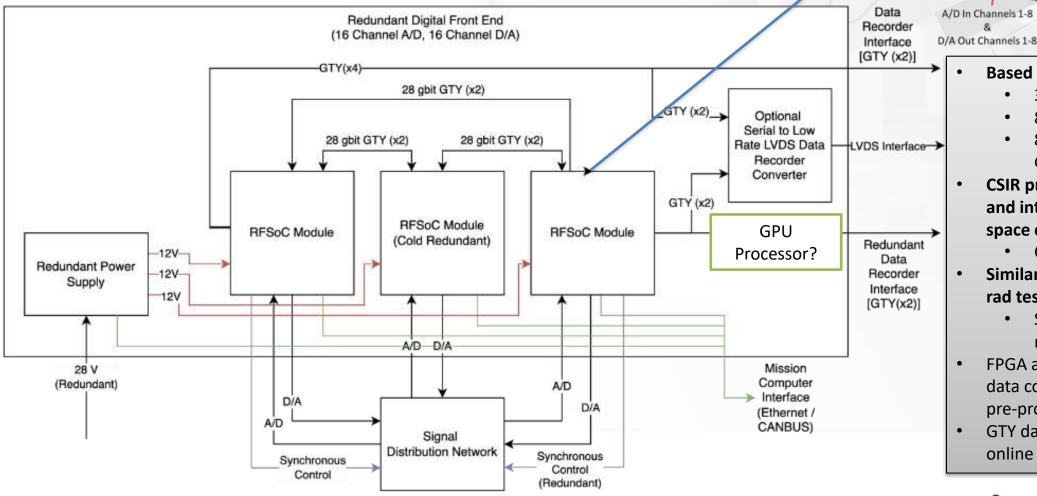


Regular Airborne SAR Flight Campaigns





## **Onboard Processing** (Digital Front End Redundancy)



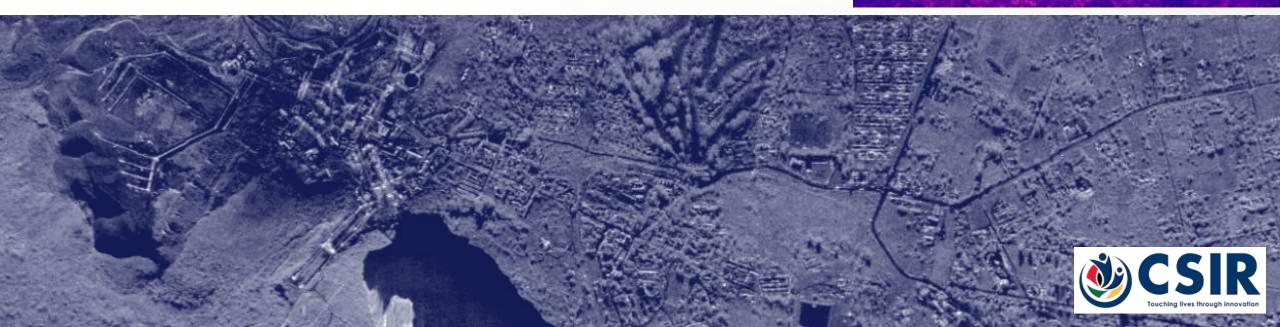
- Based on COTS RF SoC Modules
  3<sup>rd</sup> Gen
  - 8 x 14bit A/Ds (up to 5 GSPS)
  - 8 x 14 bit D/As (up to 10 GSPS)
  - CSIR provides firmware, software and integration and testing for space environment
    - Cold redundant spare unit
- Similar modules (Gen 1) has been rad tested
  - Supplier willing to fix issues resulting from rad testing
- FPGA and Arm Processor allows for data compression and first level of pre-processing (equalization, etc)
- GTY data paths allow for parallel online processor development



- **Back-projection** and more traditional SAR algorithms (Chirpscaling, RD, etc)
- Allows for advanced modes including Long Dwell and VideoSAR
- Can handle extreme squint angles
- Proven in airborne radar scenarios where requirements are more stringent due to unknown platform motion
- Data **outputs in industry standard formats** (including cooperation with industry players) to allow integration into industry workflows
- **GPU acceleration** to reduce processing lag can already process fine resolution airborne SAR data in real-time
- Allows for cluster / cloud-based compute scaling to enable constellation scaling
- Development of processing for advanced modes and constellation level processing forms part of HCD programme to train a next generation
- Grow towards on-board real-time processing through the program

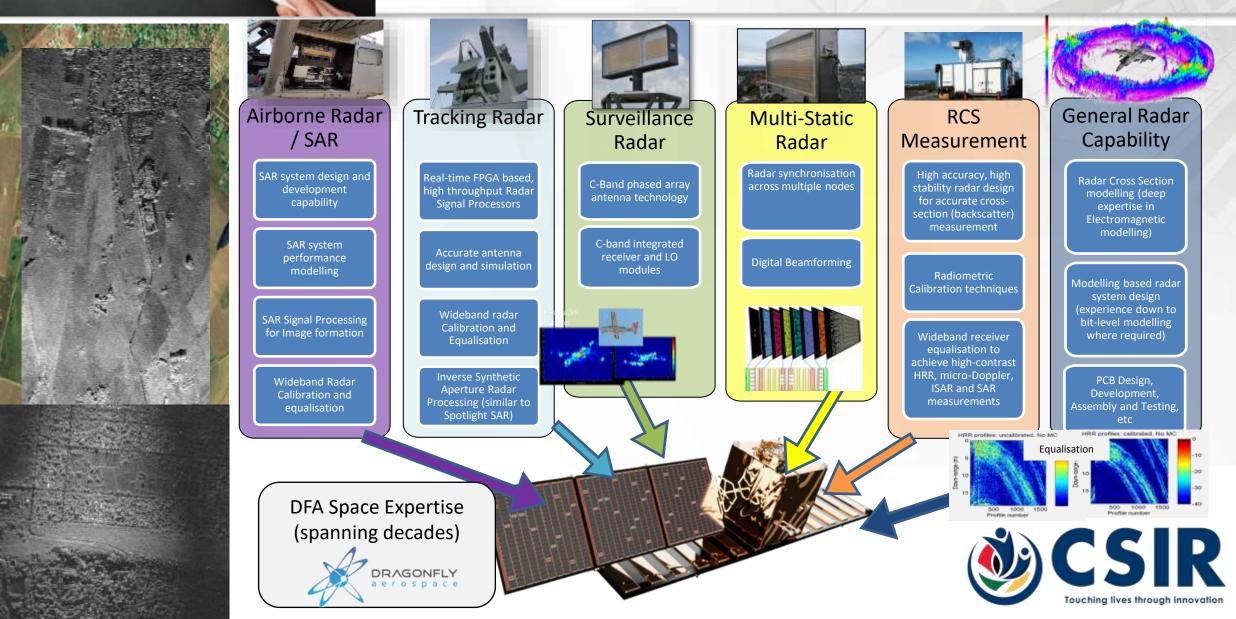
#### SAR Processor





# QUESTIONS

#### CSIR Radar (and DFA Space) know-how spans decades and is applied to the SAR-C design



## Thank you

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