



Cape
Peninsula
University
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creating futures

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CPUT's Satellite Programme.

**One
SMART
CPUT**



30 August to 01 September 2023

ESRIGC, Tshwane, Pretoria, South Africa

VISION
**20
30**

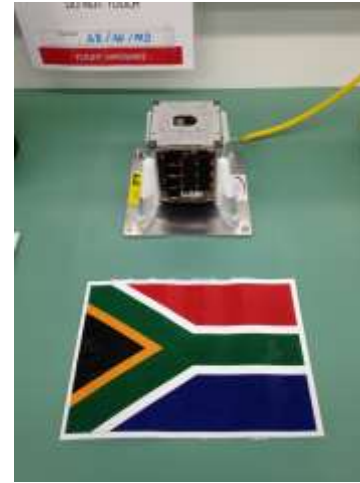
African Space Innovation Centre (ASIC) at CPUT “niche area” is the nanosatellite engineering technology

CPUT utilised this centre as post graduate training facility in satellite engineering.

Successfully demonstrated with the first SA’s student nano satellite mission:

ZaCube 1: Tshepiso Sat
(launched Nov 2013)

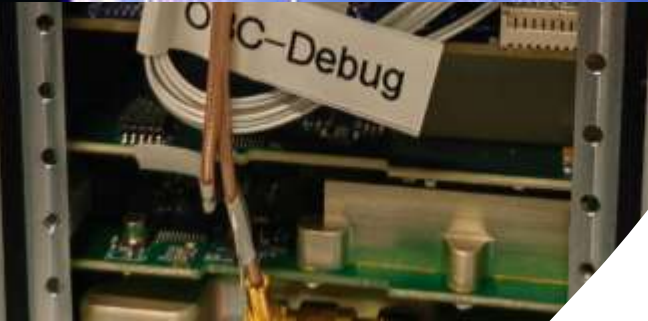
- Primary payload: HF Beacon
- Hosted payload: Visible Spectrum Imager



ZaCube-1 payload descriptions.



ZACUBE-1
cput.ac.za/fsati

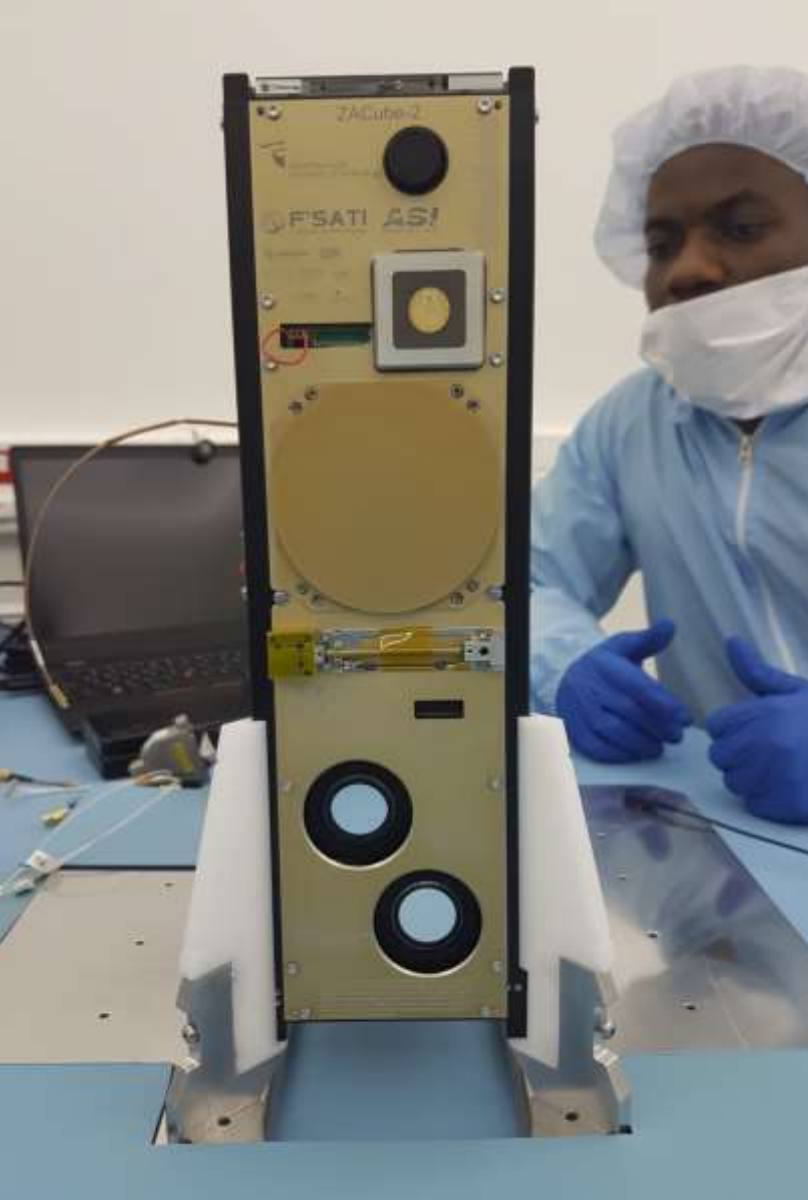


Zacube-1 – Primary payload.

- SANSA (HMO) operates an HF space weather radar at the SANAE base in Antarctica (Dual Auroral Radar Network).
- 16 element antenna array to transmit signals into the ionosphere.
- Analyse ionospheric density over time
- The whole antenna array is hundreds of meters in size and very difficult to characterise on the ground.
- One solution to characterising the array was to emit a beacon signal from a satellite and measure the signal received by the array.

ZaCube-1 – Secondary payload

- The imaging payload captured images of the earth's surface and stores these images on-board in a digital format for download.

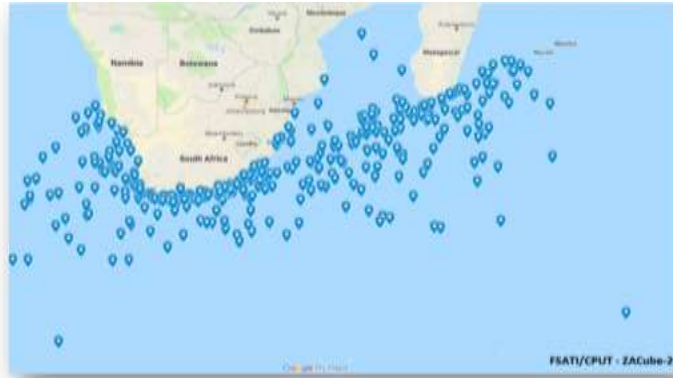
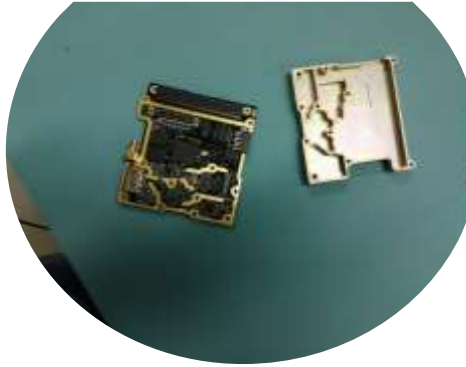


South African Oceans Economy through ZaCube-2 Demo mission

- Operations Phakisa (Maritime Domain Awareness)
 - Operation Phakisa is an initiative of the South African government.
 - It is a fast results delivery programme that was launched in July 2014 to help implement the National Development Plan.
 - It is a cross-sector programme where various stakeholders engage to implement initiatives that would enhance the development of SA's economy.
 - Oceans economy was identified as one of these initiatives.
 - A need was identified under Operation Phakisa's Marine protection services and governance (MPSG) focus area.
 - MPSG Lab developed 10 key initiatives, one of which is the National Ocean and coastal information system (OCIMS) and as the extension of earth observation capacity.

ZaCube-2 Payload Technology Demonstration

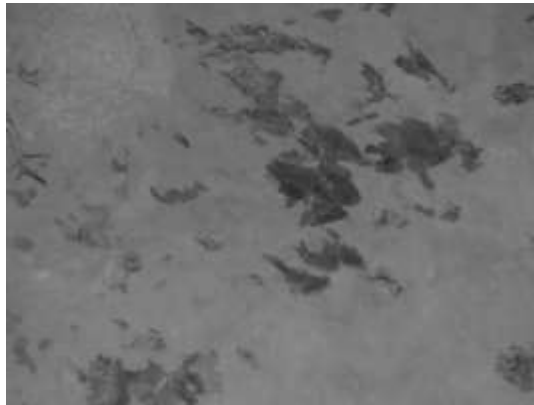
ZaCube-2: Launched 27 Dec 2018



science & innovation
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ZaCube-2 mission objectives were re-aligned and augmented to allow the satellite to function as a precursor mission for the Operation Phakisa “MDASat” constellation. ZACUBE-2 was utilised as a technology demonstrator for essential subsystems required in operational nanosatellite missions.

- Primary payload - AIS capability utilising an SDR.
- Software defined radio utilized to receive AIS messages from Ships
- Over SA’s exclusive economic zone.
- Hosted payload – K-line Near Infra-Red Imager demo.
- Vegetational fire detection
- Imager operates in part of the light spectrum where Potassium emissions are found and be detected.

Maritime Domain Awareness Satellites (MDASat-1)

MDASat was built on the heritage of ZACube-2 to demonstrate an expanded capability and data delivery.

- One of the goals of this constellation was to supply data to the OCIMS.

The first 3 satellites of the MDA constellations carried an improved AIS payload compared to ZACube-2, with these additional functionalities and improvements:

- Over-the-air upgrades,
- Capture raw data ,
- Long Range AIS
- Improved efficient messaging scheme.



Maritime Domain Awareness Satellites (MDASat-1)

Early Operations (Jan and Feb 2022)
Commissioning of MDASAT-1 Complete
Current Operation MDASAT- 1



mcs APP 10:09 AM

Rows downloaded this overpass:

Data Desc	ID	down	left	done
wod	24	211	0	0
wod	25	0	0	1
event	28	5	0	0
adcsa	56	0	0	1
adcsa	57	158	0	0

Total size of data downloaded:
35807 bytes (35.807 kBytes)

MDASat-1c overpass completed at 2023-08-30 08:09:51 (Max ele 47.8)

mcs APP 9:55 AM

Rows downloaded this overpass:

Data Desc	ID	down	left	done
wod	24	212	0	0
wod	25	0	0	1
event	28	379	0	0
adcsa	56	0	0	1
adcsa	57	159	0	0

Total size of data downloaded:
40482 bytes (40.482 kBytes)

MDASat-1a overpass completed at 2023-08-30 07:55:00 (Max ele 32.4)

Current study of the constellation and lessons learnt from the deployment

M2MSat- (VDES) Demonstrator

- Demonstrate machine to machine message transmission (i.e. ship-to-ship).
- Demonstrate machine to hub message transmission (i.e. ship-to-shore).
- Demonstrate hub to machine message transmission (i.e. shore-to-ship).
- Reception of AIS messages in the AIS1, AIS2 and LR-AIS channels.

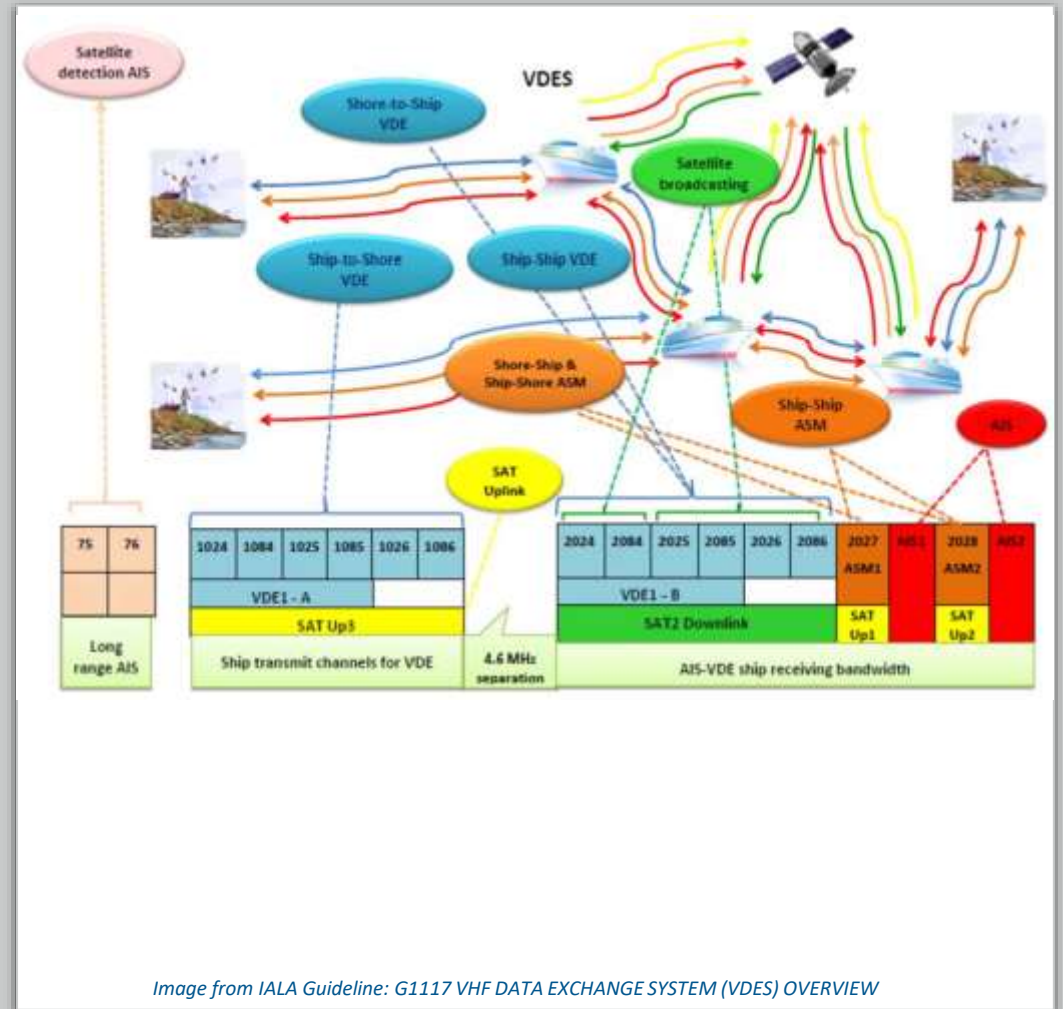


Image from IALA Guideline: G1117 VHF DATA EXCHANGE SYSTEM (VDES) OVERVIEW

M2MSat-1 Development progress

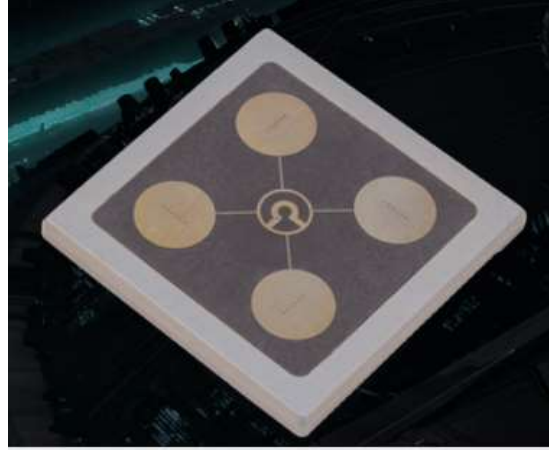
Soft-stack and EQM Integration



CubeSat Communication Subsystems

S-Band and X-Band Antennae

Off-the-shelf nanosatellite products generated from the ASIC programme



General	XANT	XANT-PLUS	General	
Frequency Range	8.025 GHz - 8.45 GHz	8.025 GHz - 8.45 GHz	Frequency Range	2.2 - 2.3 GHz (SANTC) 2.4 - 2.45 GHz (SANT)
Bandwidth	200MHz	420MHz	Beamwidth	60°
Temperature	-40°C to +85°C	-40°C to +85°C	Temperature	-40°C to +85°C
Mass	< 9 g	< 29 g	Mass	< 50 g
Gain	7.75 dBi ± 0.5 dB	11.5 dBi ± 0.5 dB	Gain	7 dBi
Axis. Rate	< 3dB across the band	< 2 dB across the band	S ¹¹	< -15 dB
S ¹¹	< -15 dB	< -14 dB	Connector	SMA female
Polarisation	LHCP / RHCP	LHCP / RHCP	Design Life	> 2 years in LED
Max RF Power	3W (33 dBm)	3W (37 dBm)	Dimensions	1U form factor
Connector Type	Straight SMP	Straight SMP		
Design Life	2 years in LED	2 years in LED		
Width	36 mm	58 mm		
Height	4.7 mm	4.7 mm		
Length	36 mm	58 mm		

CubeSat Communication Subsystems

VHF/UHF and S-band Radios



General	
Operating Temperature	-25°C to +51°C
Mass	< 100 g
Voltage	3.3 V, 5 V
Frequency	
UHF	400 - 420 MHz (UTRXC) 430 - 440 MHz (UTRX)
Transmit	
DC Power	3 - 0.1 W (27-13 dBm)
RF Power	27 - 33 dBm (12 dB slope)
Channel Spacing	25 kHz
Spurious Response	< -45 dBc
Frequency Deviation	± 4 kHz (FM)
Frequency Stability	± 3.3 ppm
Receive	
DC Power	< 240 mW
Sensitivity	-115 dBm (for 12 dB SINAD)
Channel Spacing	32.5 kHz
Noise Figure	< 2.5 dB
Dynamic Range	-115 to -70 dBm
Frequency Stability	± 2.5 ppm

Performance	
Processing	<ul style="list-style-type: none"> Low-power Flash-based FPGA CRC-16-DCTT (AX.23) Scrambling (BMSK) Transparent datalink mode 1/2 Rate CCSDS convolutional encoding (w/1) available in transparent mode
Interfaces	<ul style="list-style-type: none"> DC Bus - 400 kHz telemetry, command and user data Receive (ready) output line Transmit (ready) output line
Modulation & Protocol	<ul style="list-style-type: none"> GMSK (FSK based) AFSK (FSK based) AX.25 Protocol Transparent mode
Dimensions	
Length	96 mm
Width	90 mm
Height*	16.9 mm

*Height from top of enclosure to lowest component on bottom.

General	
Temperature	-25°C to +51°C
Power	< 5 W
Voltage	6 V - 12 V (5 V alternative)
Mass	< 100 g
TX SNR	20 dB
Frequency	2.2 GHz - 2.3 GHz (PULSAR-STXC) 2.4 GHz - 2.45 GHz (PULSAR-STX)
RF Power	24 - 30 dBm (1 W RF) in 2 dB steps
Channel Spacing	500 kHz
Spurious Response	< -40 dBc
Design Life	2 years in LEO

Performance	
Processing	<ul style="list-style-type: none"> Low-power Flash-based FPGA V.35 IntraSAT scrambler 1/2 rate convolutional encoding (K=7) Differential encoding Pulse shaping filter
Interfaces	<ul style="list-style-type: none"> Low-speed I2C Bus - 400 kHz (telemetry and control) High-speed SPI Bus - 4 MHz (payload data) 50 Ω SMA connector
Modulation	<ul style="list-style-type: none"> QPSK or OQPSK InterSAT IESS-30B

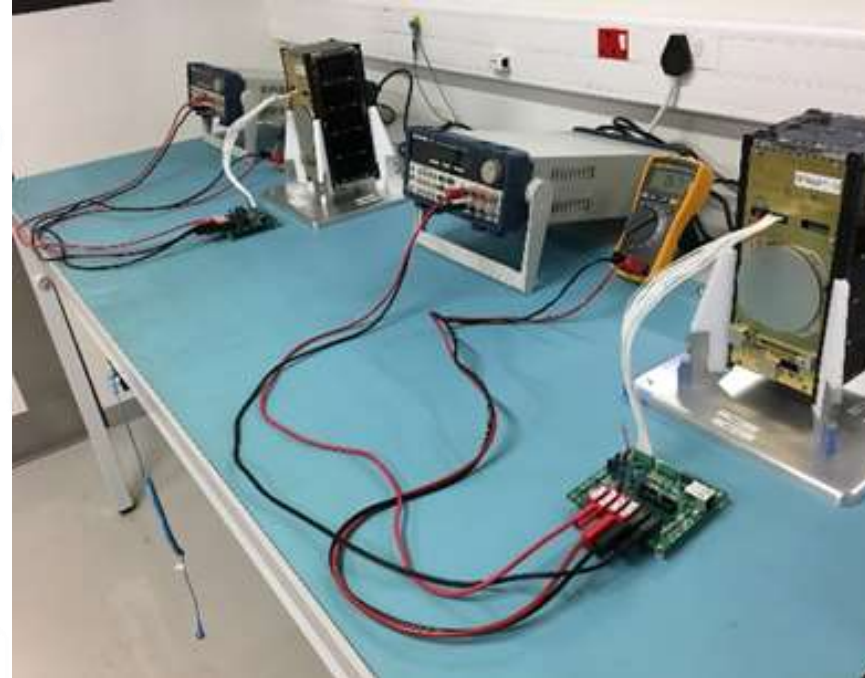
Dimensions	
Length	96 mm
Width	90 mm
Height	16.9 mm

*Height from top PCB to lowest component

Engineering Research Facilities



(1) Production area



(2) FM area

Engineering Research Facilities



(1) Clean room

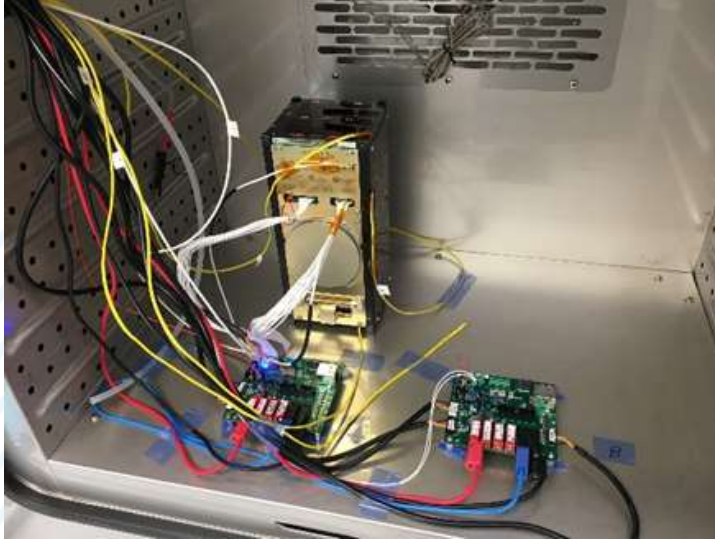


(2) Ground station

Supports:

- VHF, UHF, and S-band
- 1.2kbps AFSK and 9.6kbps GMSK uplink or downlink.

Engineering Research Facilities



(1) Thermal cycle chamber



(2) Workspaces and classrooms

Conclusion

The satellite missions presented here are displaying progress towards developing a sovereign capability in space technology development and educational training for developing countries like South Africa.



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**THANK YOU
BAIE DANKIE
ENKOSI**